







Control Systems for Thermal Vacuum Chambers Benjamin Chu

NASA Space Grant Caltech Student-Faculty Programs

Thermal Vacuum Chambers JPL

- Flight hardware needs to be tested in a thermal vacuum chamber to mimic the vacuum and temperature of space.
- Some tests conducted at the Reliability Assurance Technology Test Laboratory (RATT Lab).



Cassini in the 25 ft Space Simulator.







- Chamber currently used to develop instruments.
- A vacuum fail-safe device is to be built for one of the chambers to increase the margin of safety.
- Installation of thermal actuators.
- Goal of flight certification.



Vacuum Systems



- Knudsen Number (Kn): mean free path of molecules divided by pipe radius.
- "Rough" vacuum:
 - Up to 10⁻³ Torr (mm Hg), Kn<0.01.
 - Air is viscous and pulled with a pressure difference (rough pump).
- "High" vacuum:
 - up to 10⁻¹⁰ Torr (mm Hg), Kn>1.
 - Air acts like particles and pushed with momentum transfer (turbomolecular pump).



Turbo pump.



Heat Exchanger Systems

- Heat exchanger plate used to change the temperature of the chamber.
- Resistance heaters and cryogens (liquid nitrogen, LN₂) used for heating and cooling.



LN₂ dewar flasks that supply the RATT Lab.





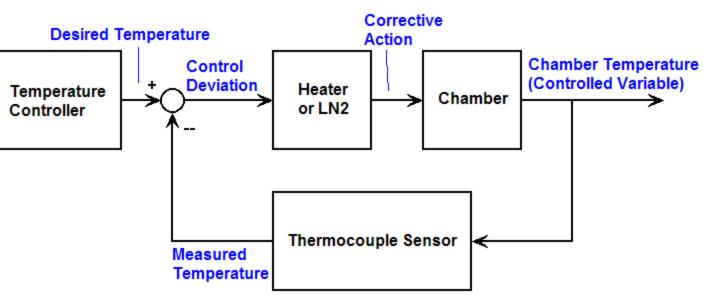


- Thermocouples monitor temperature.
- Thermocouple gauges to measure the vacuum up to "rough" vacuum (10⁻³ Torr).
- Bayard-Alpert ionization gauges for up to "high" vacuums (10⁻¹⁰ Torr).
- A multi-gauge controller is used to measure the output of the two vacuum gauges.



Bayard-Alpert ion gauge.

Control Systems



- Control systems maintain a constant vacuum and temperature environment by integrating sensors and actuators.
- Overall system consists of three components:
 - Temperature controller: maintain preset temperature.
 - Temperature fail-safe: backup for extreme temperatures.
 - Vacuum fail-safe: backup for loss of vacuum.

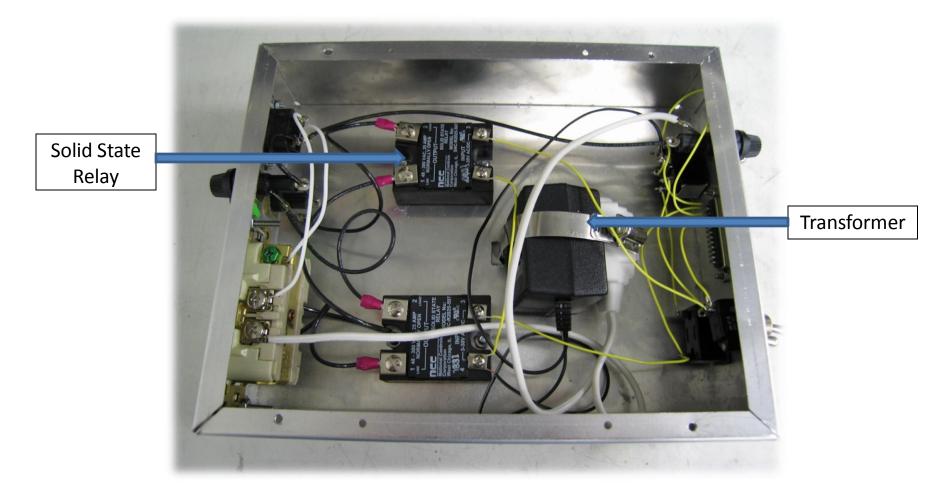


Fail-safe Mechanisms JPL

- Temperature fail-safe:
 - High temperatures: heaters shut off.
 - Cold temperatures: solenoid valves shut off the flow of LN₂.
- Vacuum fail-safe:
 - Loss of vacuum: isolate the chamber to protect the instruments and close foreline to protect turbo pump.
- Actuators and sensors to be installed for the temperature controller and temperature fail-safe.
- Vacuum fail-safe needs to be built.



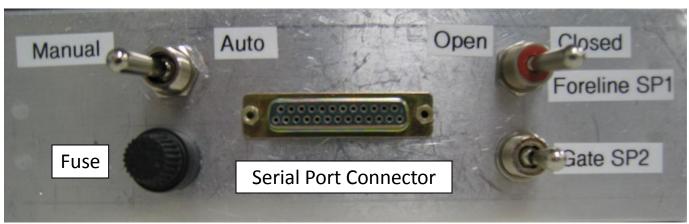
Vacuum Fail-safe JPL



Inside of the vacuum fail-safe box.



Vacuum Fail-safe



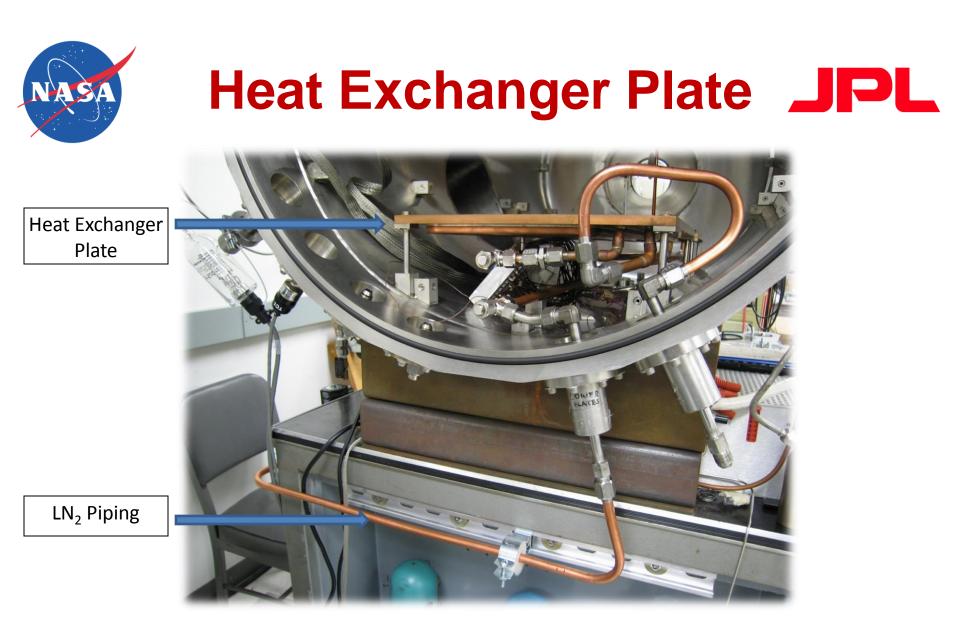
Front view of the vacuum fail-safe box.



Back view of the vacuum fail-safe box.



Solenoid valve setup. One valve controlled by controller, one by fail-safe.



LN₂ and resistance heaters for cooling and heating inside the chamber.



The control and fail-safe units.



Results



- Vacuum leak simulated to test the vacuum fail-safe.
- Fail-safe correctly latched on to "safe" mode.
- Thermocouples connected to data acquisition device to record temperature during test.
- Simulated hot and cold temperatures.

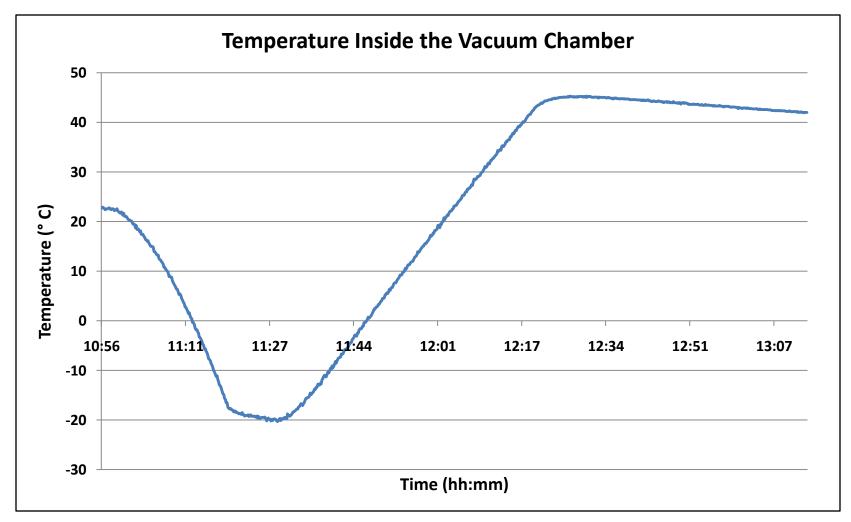


The data acquisition system.



Temperature Test





Graph of temperature over time. The fail safe was set to -20°C and 40°C.







- Vacuum and temperature fail-safes work.
- Some overshoot of temperature, but negligible:
 - Location of the thermocouples.
 - Thermal inertial of heat exchanger plate.
 - Cold gas remaining in piping.
- Updated existing operating procedures to include procedures for the vacuum fail-safe.



Acknowledgements JPL

- Caltech Student-Faculty Programs and NASA Space Grant.
- Paul Bowerman and Nelson Green.